



MiniMill: a miniature Field Mill Electrometer for airborne platforms

Img: NASA Mars Reconnaissance
Orbiter



FOSDEM'24

***Vasiliki Daskalopoulou,
V. Spanakis and Th. Georgiou***

Agenda



Intro – Scientific Question

Why open source ?

Sensor Assembly & Testing

Observations

Data sharing

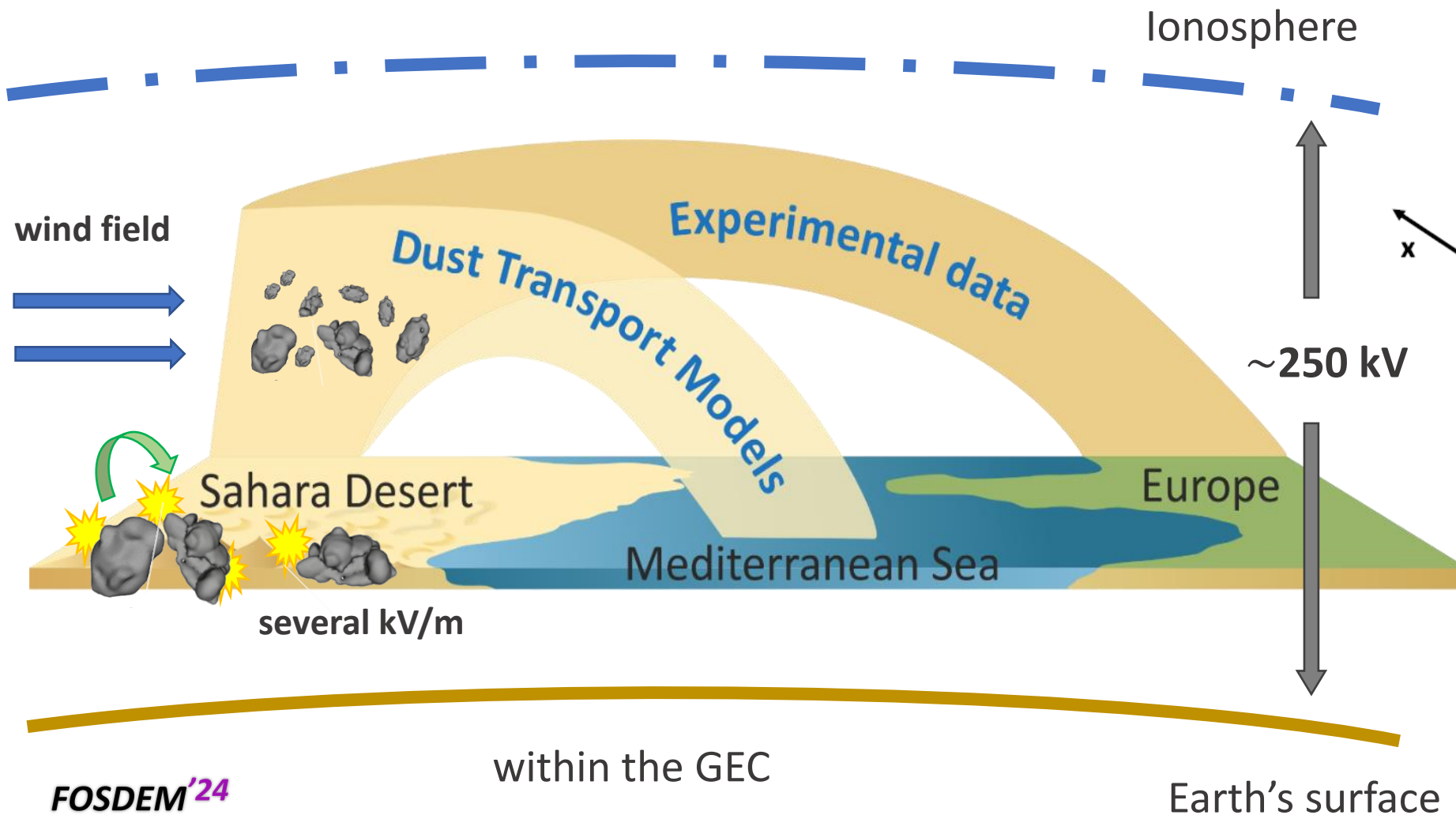
What next ?



Impact of Atmospheric Electricity on particle dynamics



from Saltation to long-range transport



Ionosphere

~ 250 kV

$\vec{F}_{electric}$

1

$\vec{F}_{buoyancy}$

\vec{F}_{drag}

x

y

θ

a

b

CG

~~$\vec{F}_{gravity}$~~

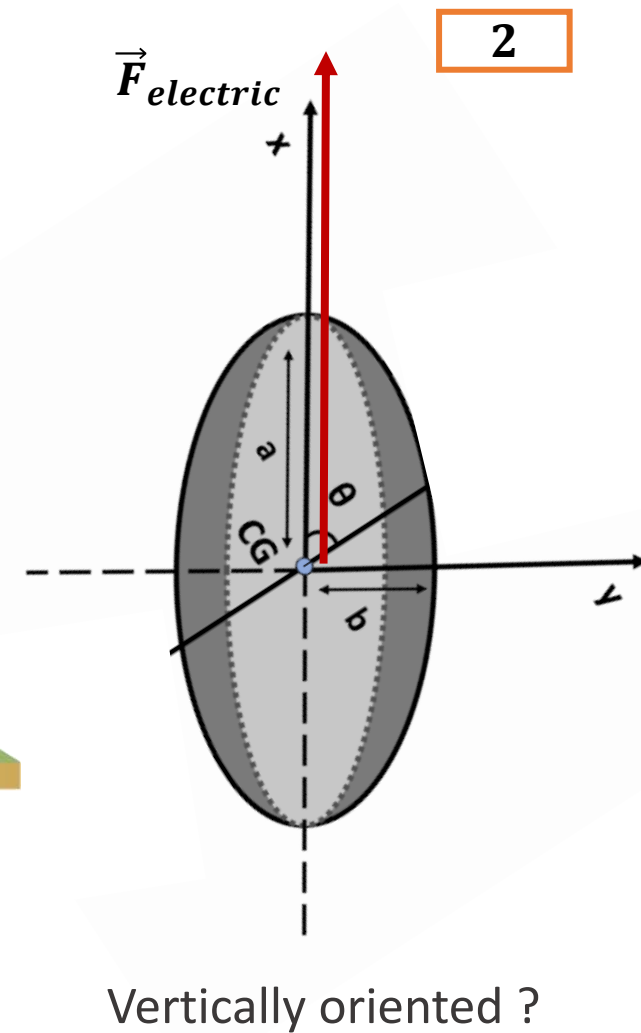
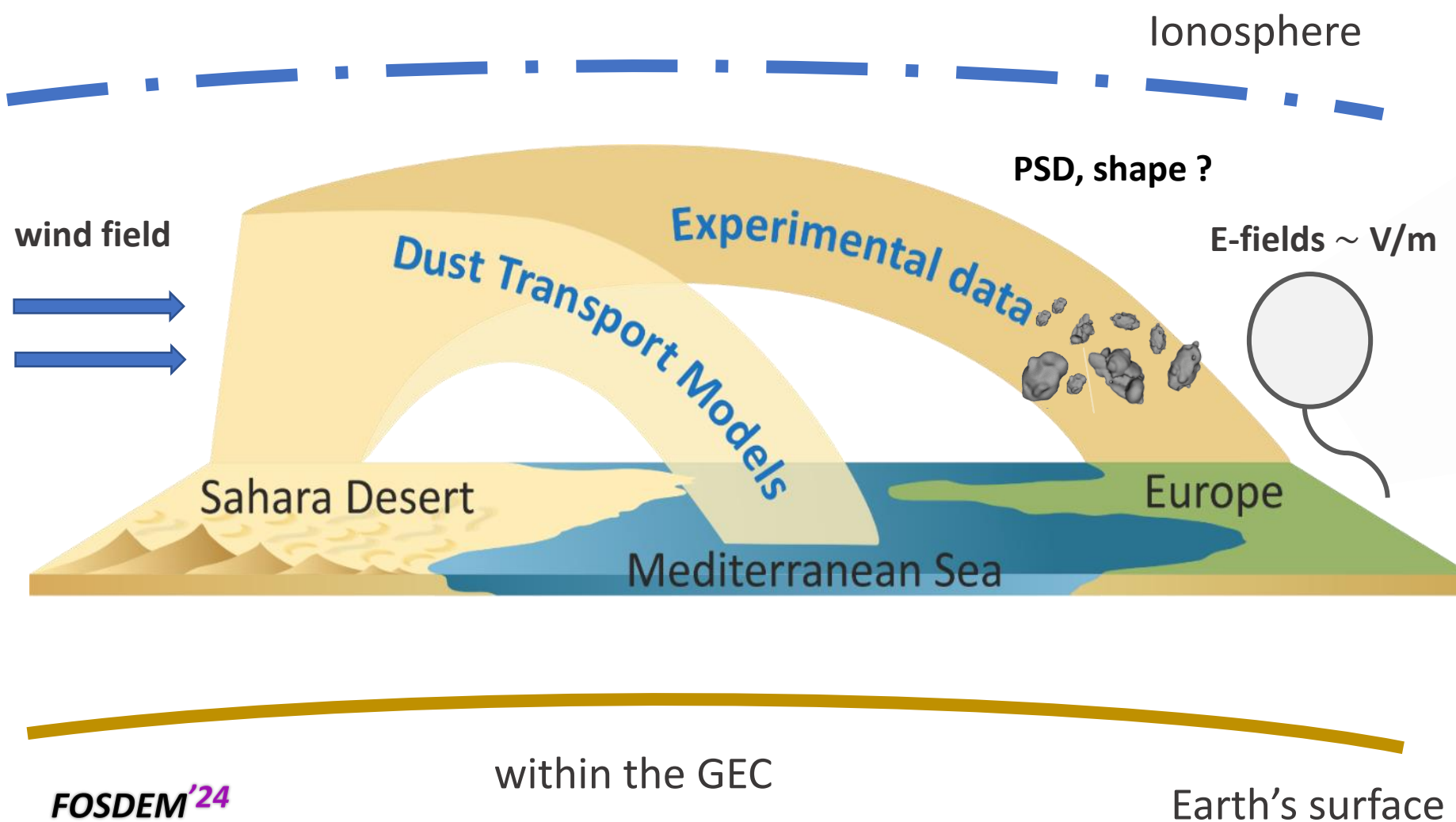
Negated ?



Impact of Atmospheric Electricity on particle dynamics



from Saltation to long-range transport

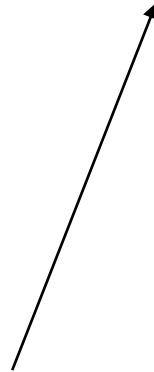




New developments



Can we verify our hypothesis through observations ?



Vertical **profiling** of the electrical properties

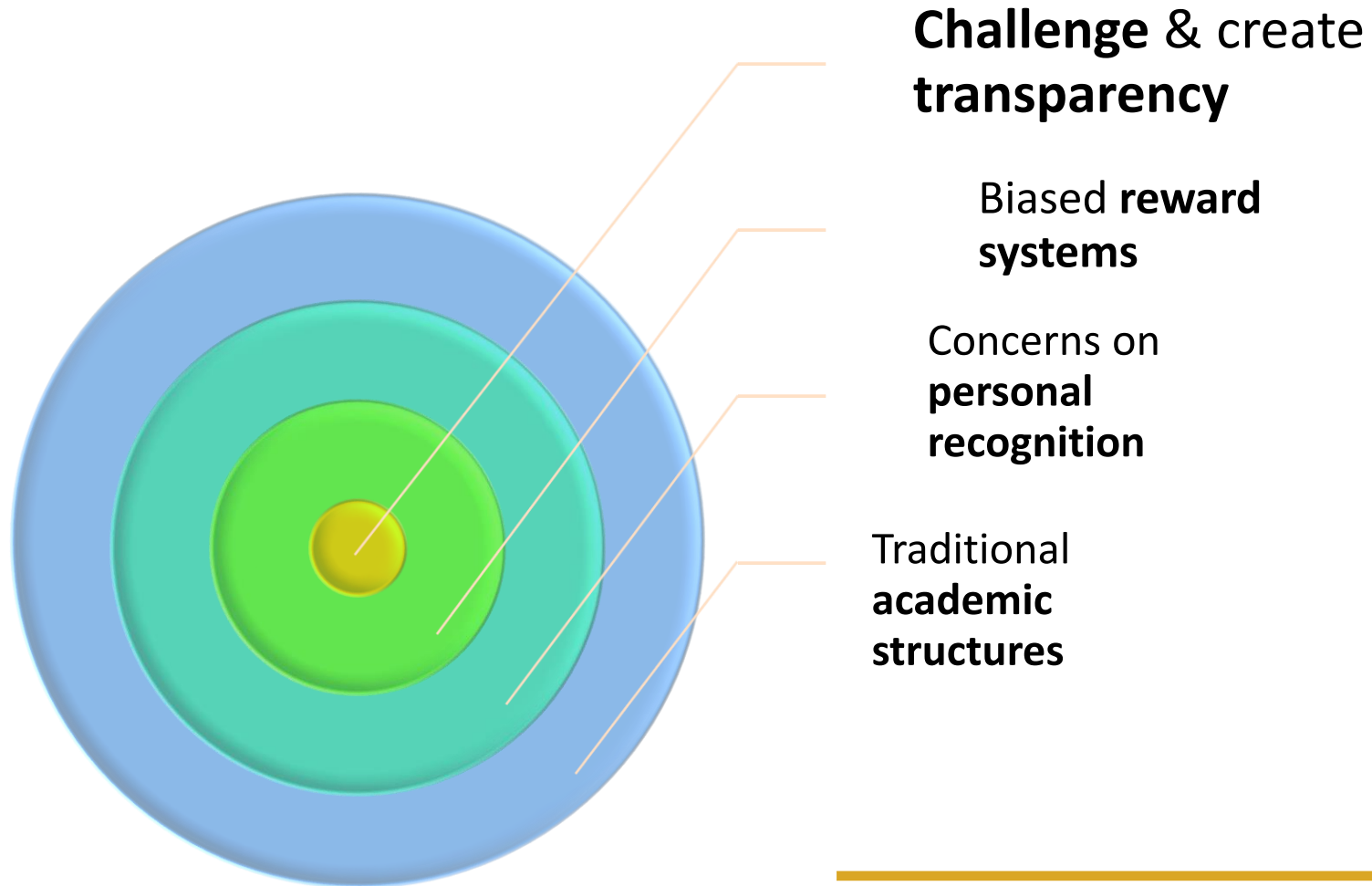
*“Hope clouds
observation.”*

Frank Herbert,

Dune



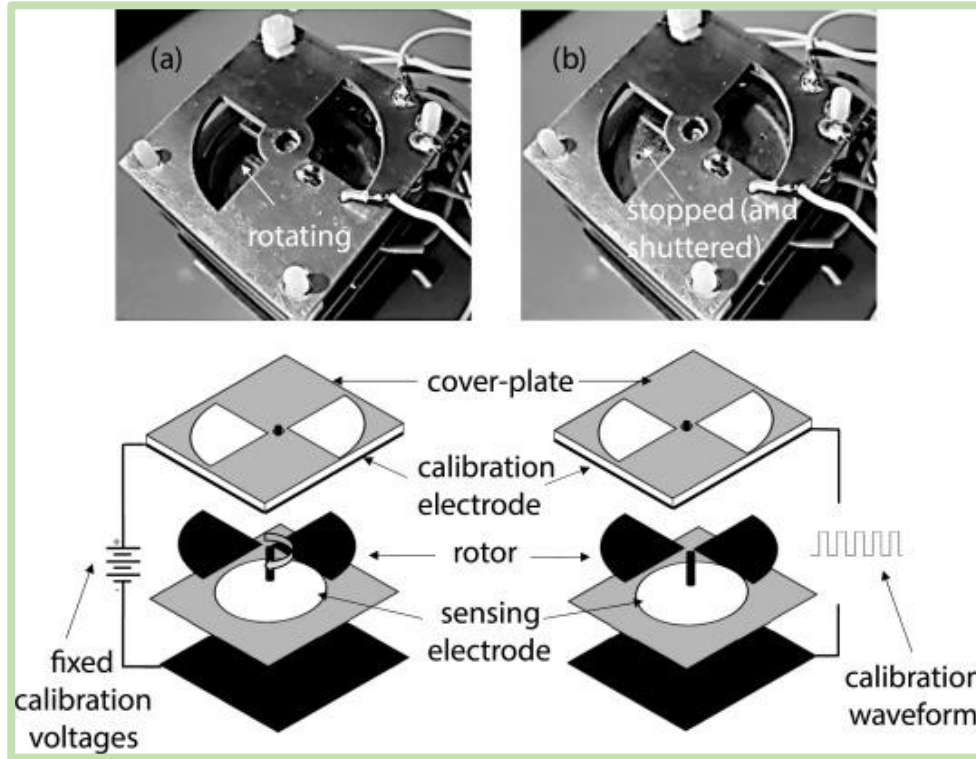
Why open-source ?



What we found up till then was fairly **closed** ...



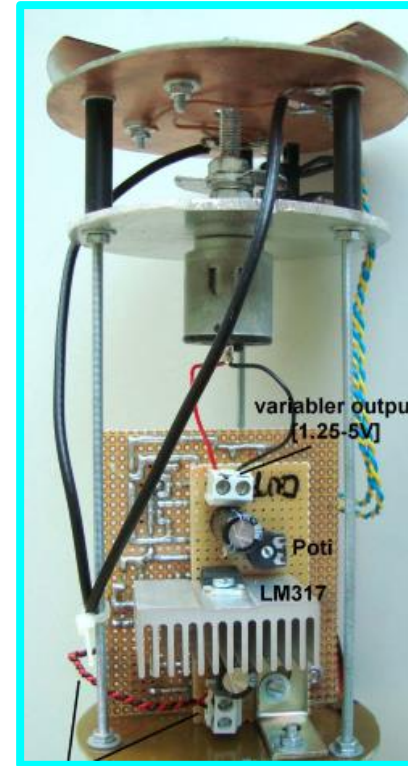
Different implementations of miniature fieldmills



Homebrewed stuff

- Non portable

Not sensitive to
Fair weather fields



Cool project

@ArcAttack/FieldMill-PCB

- Not tested
balloon-borne
platforms

- Sketchy or non-existent schematics

Harrison et al., 2020

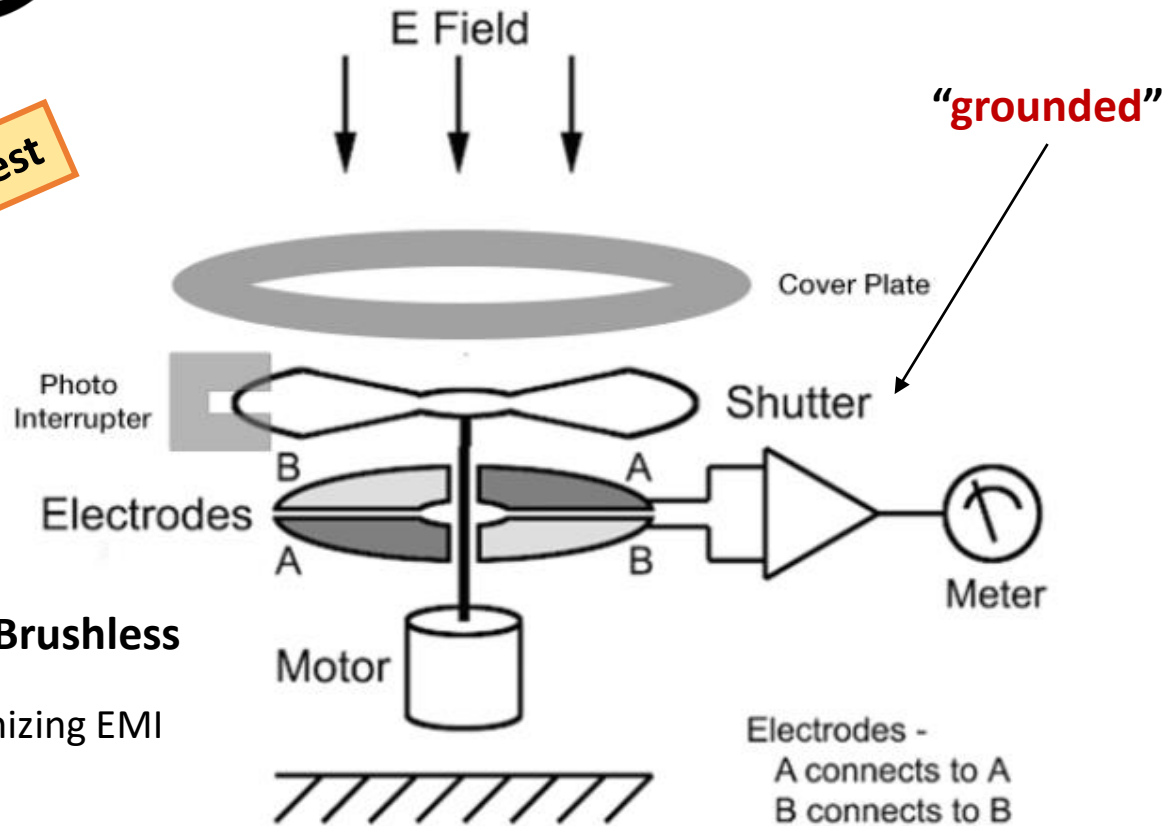
& Cui et al., 2017



MiniMill: Principle of operation & Design



latest



DC Brushless

minimizing EMI

MPU-6050

gyro

Low rpms for optimal sensor response

- Periodical screening of a sensing electrode
- The **induced charges** for an effective sensing area $S(t)$:

$$Q(t) = \sum_S q = \epsilon_0 E S(t)$$

S depends on the circular sector angle

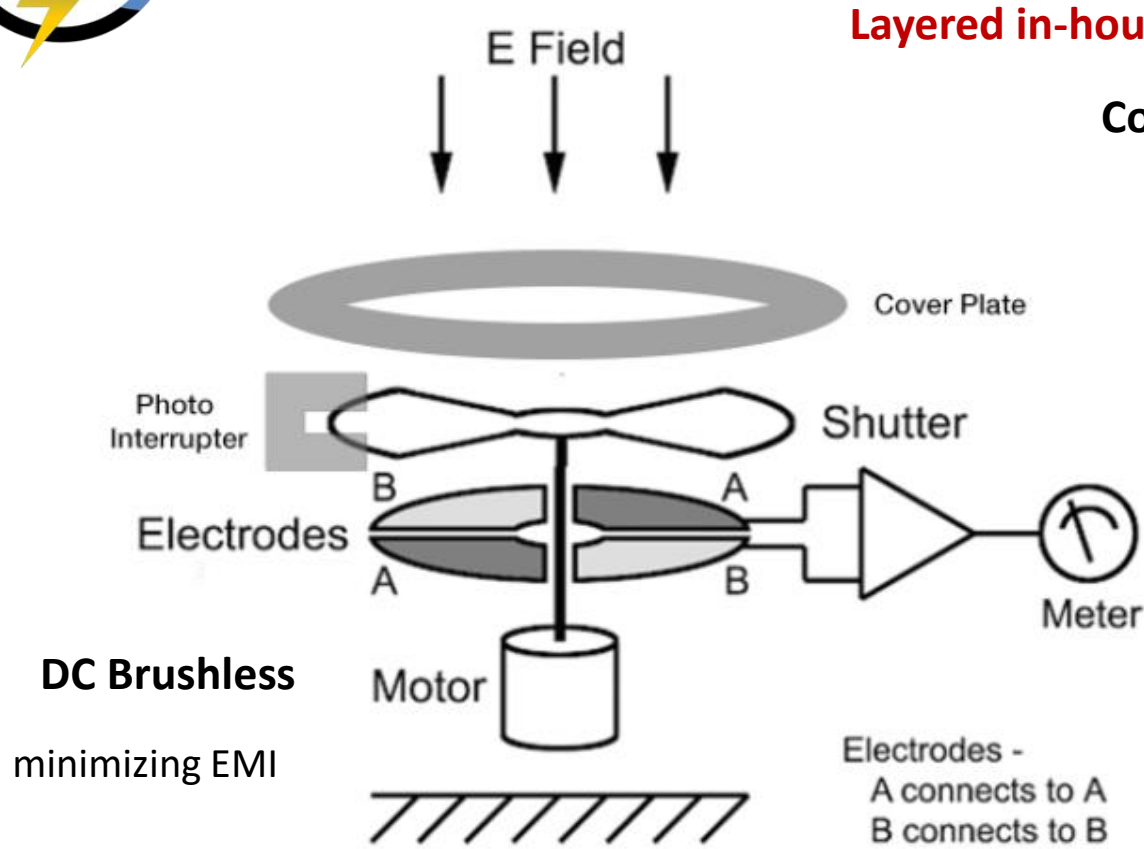
and the **induced current** directly related to the **vertical E-field** will be:

$$i(t) = \epsilon_0 E \frac{dS(t)}{dt}$$

- **Measured voltage** is twice the amplified output voltage from each electrode (differential measurement)

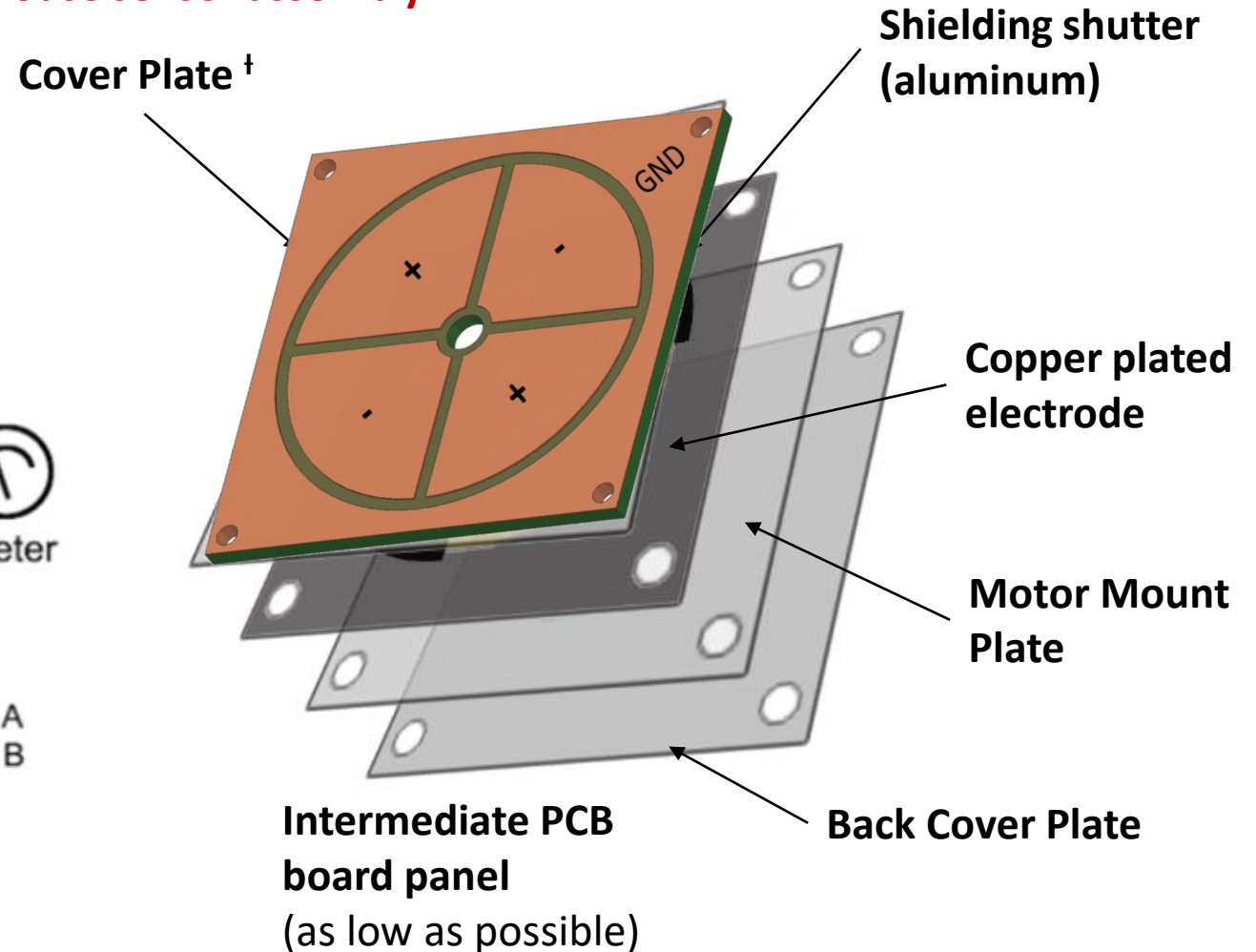


MiniMill: Principle of operation & Design



ADC microcontroller- **Serial** transmission

Layered in-house sensor assembly

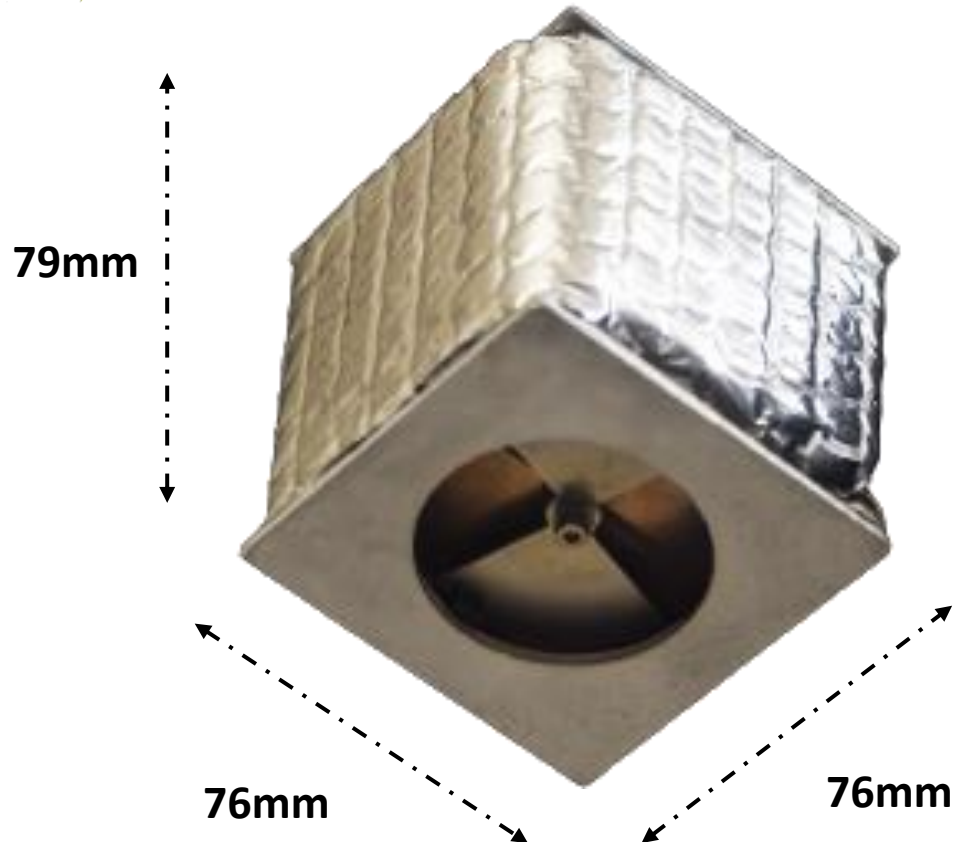




Sensor specs & limitations



With **thermal** shielding



E-field strength (V m^{-1})
(during radiosonde ascend /descend)

Pros

- ✓ Robust design; easy to reproduce; low-cost (~ 100€ each)
- ✓ Lightweight (~ 300gr) ; disposable
- ✓ Direct measurement; angular position measurements
- ✓ Sensitivity $\pm 2.3 \text{ mV per V m}^{-1}$; range $\pm 2.4 \text{ kV}$

Cons (cur. design)

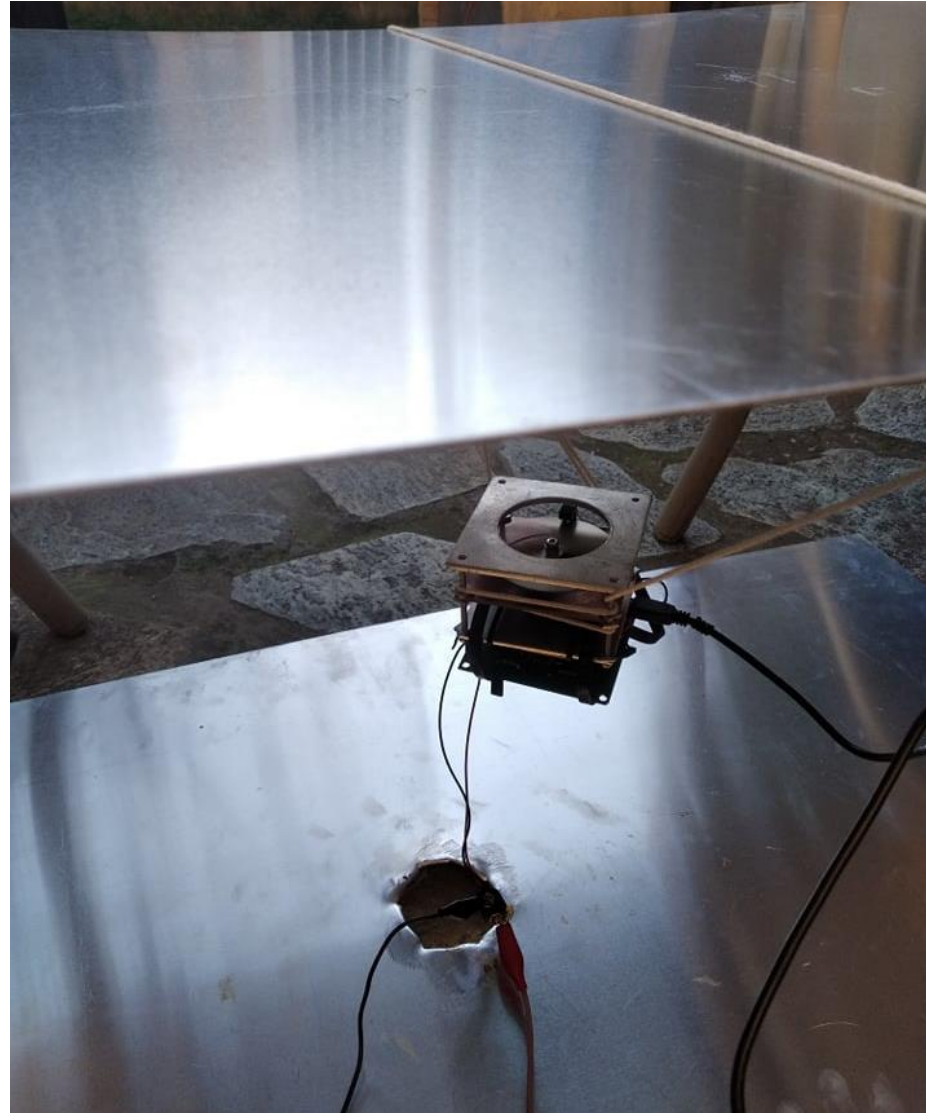
- ✗ Bulky electronics
- ✗ slightly overestimates E-field (parasitic fields, horizontal slide or $E_{x,y}$ components)
- ✗ Limited operation temperature at -50°C
- ✗ Max. altitude of operation ~ 16 Km due to motor/battery freeze

Calibration tests (completed)

i. Hard vibration test, ii. Temperature resilience, iii. FW response vs commercial Fieldmill, iv. Faraday cage response, v. parallel plates calibration & vi. test radiosonde flight



Standard calibration set-up



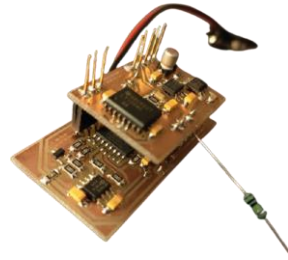
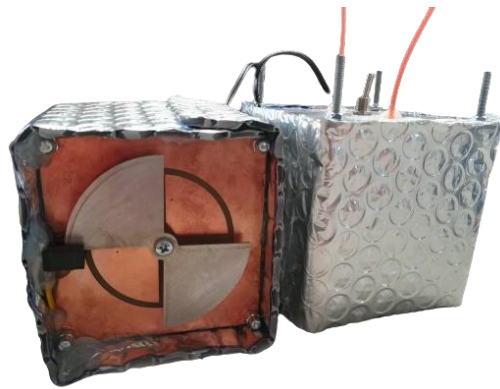
**Parallel plates with
fixed voltage input**



Telemetry through UART



Daisy chaining to GRAW meteorological radiosonde



2nd sensor pings the MiniMill



(P, T, H, U, GPS)
XDATA protocol



- Standalone on the ground:

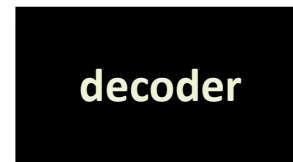
Serial to text output

Pin	Use	Comment	Connect To
1	N/C	The red wire	No connection
2	Rx		Microcontroller UART Tx
3	Tx		No connection
4	GND		Ground



“trick” GS sensor ID

- Raw data



Data storage





Mass production for experimental campaigns





Observations of Dust Electrification



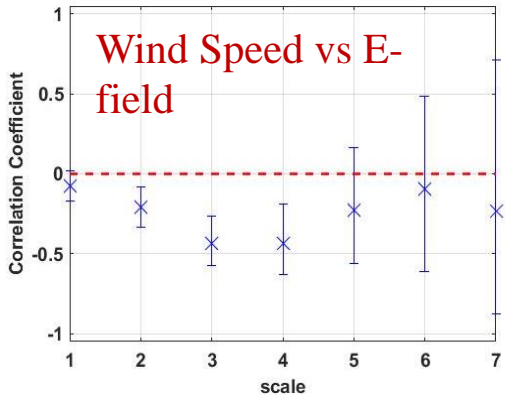
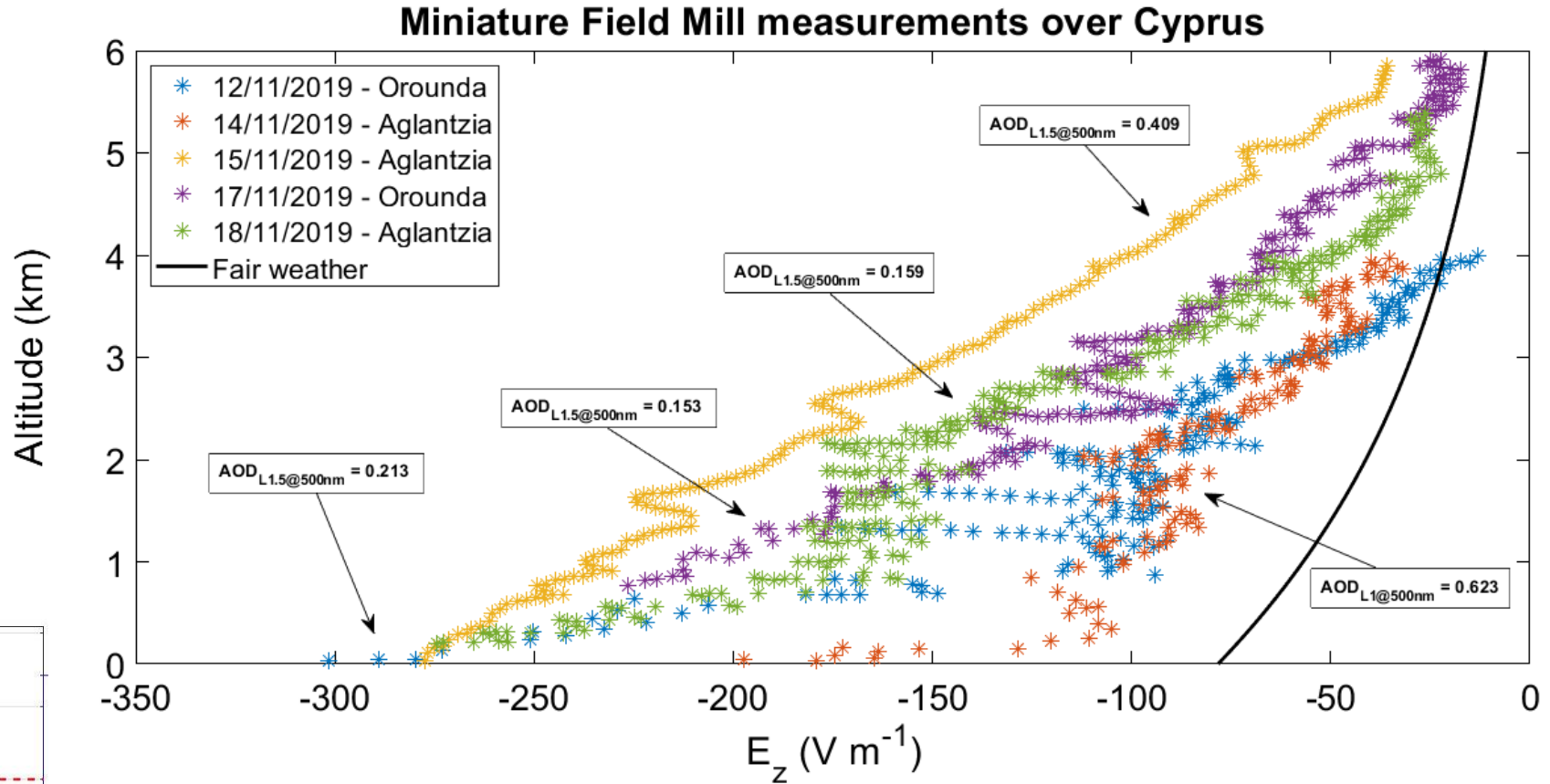
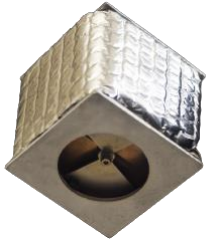
ASKOS

esa





Preparatory ASKOS experiment



Maximal overlap discrete wavelet transform

→ anticorrelation in medium scales



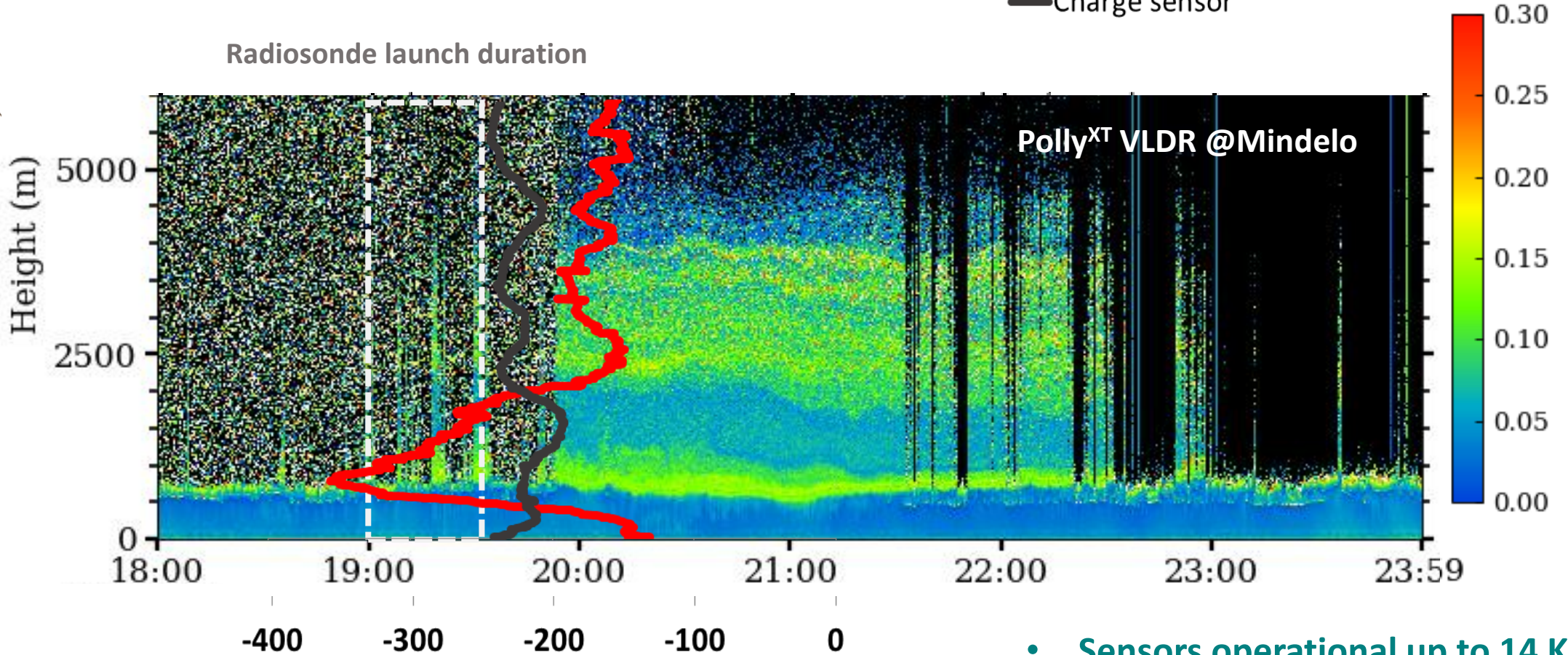
Observations of dust electrical properties



E-field for 23/06/2022

— MiniMill
— Charge sensor

Radiosonde launch duration



- Sensors operational up to 14 Km
(due to battery decay)



Data sharing platforms



Full repo: <https://github.com/NOA-ReACT/electricity-sensors>

Do contribute !



evdc
esa eo validation data centre

esa

Raw Dataset
publicly available

ASKOS

Calendar
publicly available

zenodo

open Repo



Open Research challenges



- **Standardization Issues**: lack of it may lead to variations in sensor specs, affect reliability and consistency of collected data.
- **Documentation Quality**: crucial for project success. Incomplete or unclear documentation can hinder the assembly process, especially for those not familiar with the specific hardware.
- **Technical Expertise**: for sensors assembly might be required. Ensuring that contributors or users have access to adequate support and resources for troubleshooting technical issues is crucial.
- **Data Calibration and Validation**: for regulatory compliance with research organizations
- **Funding Constraints**: often on limited budgets - testing, and QC can be a challenge, impacting the overall success.
- **Long-Term Support**: addressing hardware issues, releasing updates, and ensuring ongoing compatibility with evolving technologies.



Key takeaway



Eventually, *do we answer the sc. initial question ?*

Yes, **MiniMill** current version produced fairly nice results !

In retrospect:

Electrical Properties seem to **not play a significant role** on the long-range transport of desert dust 👎



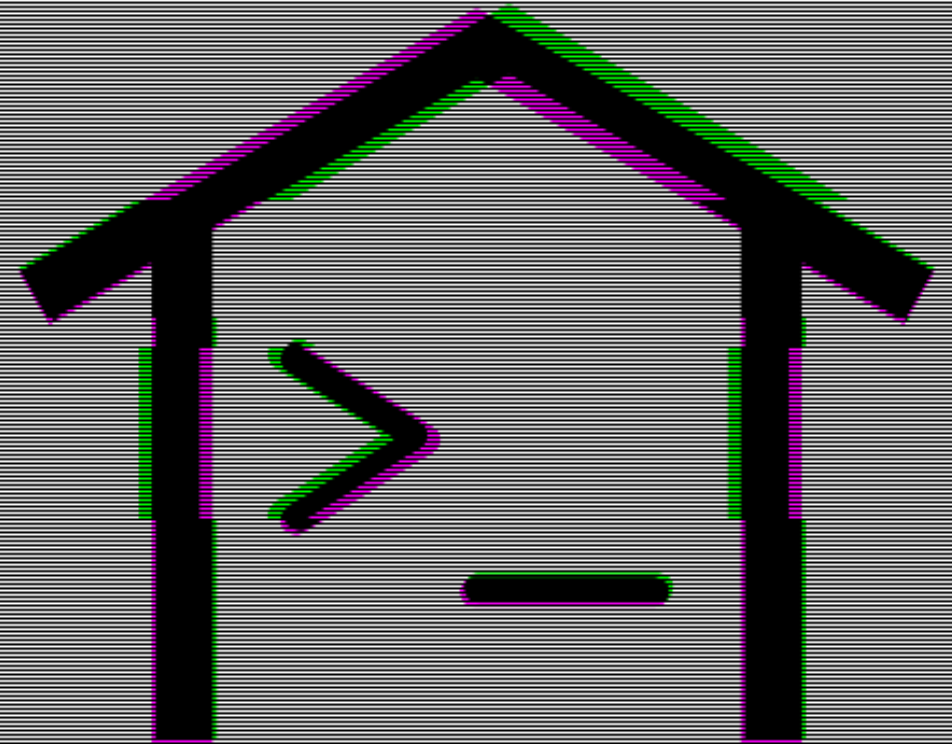
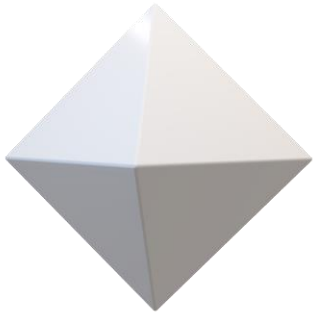
What next ?



Version 2.0

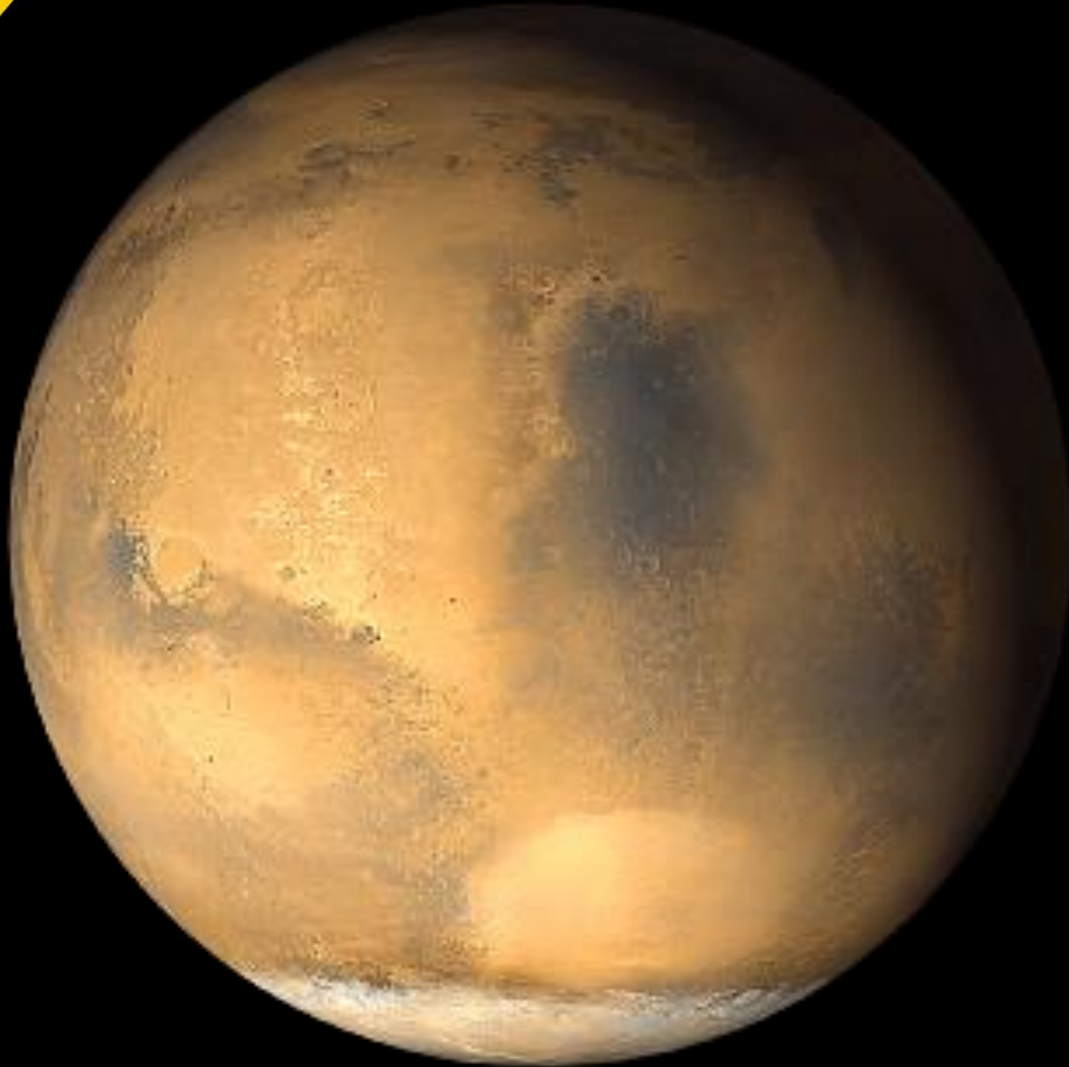


LOADING





What next ?



Thank you... all !



vdaskalop@noa.gr

[@ModusElectricus](#)



MiniMill specs



Baud Rate				Dynamic range	
Pre Radiosonde	115200			single sensitive channel	
On Radiosonde	9600	8 data bits, no parity, one stop bit (8N1)			
RPMs:				Noise	
~ 40 Hz	2400 (average)			Zero field error	~ 2mV
				Field uncertainty	± 3V/m
Sampling Rate				Sensitivity	1 ADC cnt, ± 2.3mV per V/m
Pre Radiosonde	9500 Samples Per Second			Resolution	0 - 1024 ADC cnts, ± 2.4kV
On Radiosonde	1 Sample Per Second			Accuracy	1 ADC cnt
Modes:				Bandwidth (dB)	through UART Radiosonde defined bandwidth
1) w/ Accelerometer	MPU6050 - Triple Axis Gyroscope & Accelerometer IMU			Physical	
2) No rotational information				Mass	250 gr
				Power	9V batteries (DC in)
Motor:	FlyCat 2204/260kV Brushless Motor			Cosumption	< 160mA
Speed controller:	XXD HW30A 30A Brushless Motor ESC			Electrode Deck	
PCB type:	multilayered			Radius	45mm
PCB1	PTH, 70X70mm, FR4 1mm, 35µm			Shutter	
PCB/Electrode	Electrode Deck with copper plating			Radius	5.83cm
Microcontroller type				Thickness	1mm
Arduino Nano CH340				Distance from the Electrode	4mm
Operating environment	(implemented configuration)			Cube Dimensions:	
Temperature	up to - 40°C			Height	79mm
Altitude	msl to ~ 13km			Length	76mm
Humidity	to 100%			Width	76mm

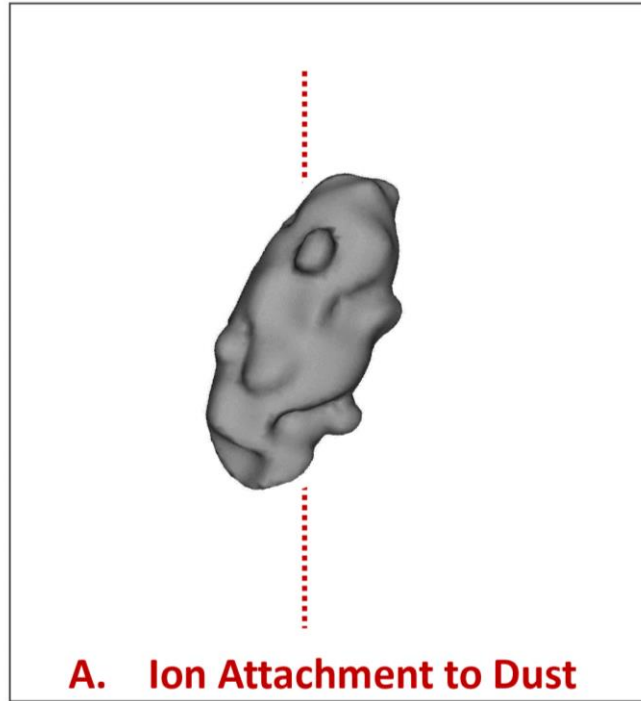


Dust Charging parametrizations

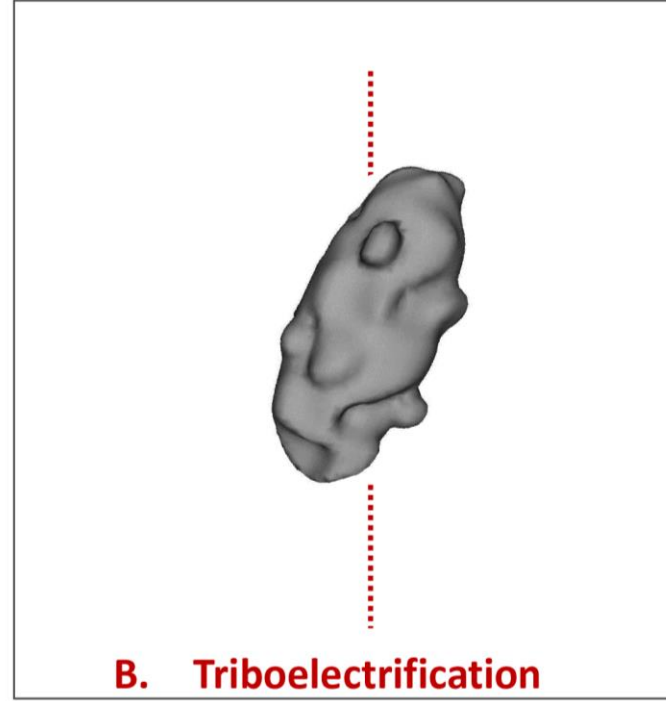


Physical Mechanisms of Dust Particle Electrification

\vec{E}



Mallios, Daskalopoulou et al., 2021



Mallios, Daskalopoulou et al., 2022

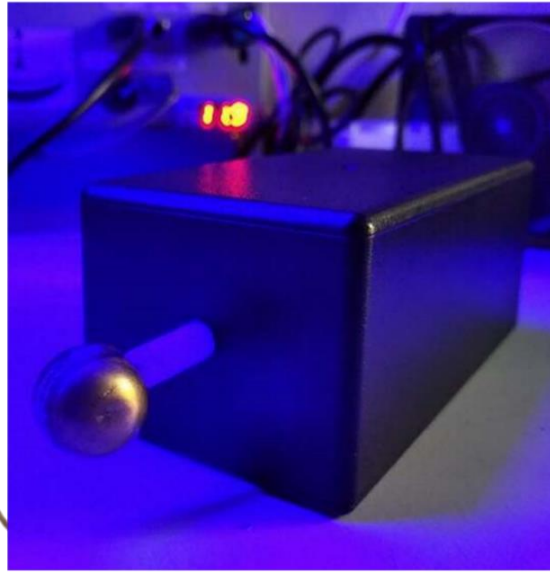
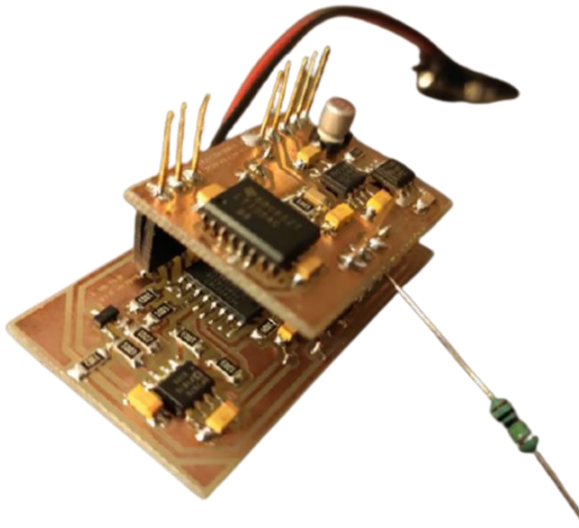
Ion attachment dominates in the case of transported dust layers



New developments for Vertical profiling: Space Charge sensor



With **hollow** brass electrode



Nicoll et al., 2013, 2011



Charge density (C m^{-3}) & **E-field strength** (V m^{-1})

(during radiosonde ascend /descend)

GitHub: @NOA-ReACT/electricity-sensors

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Pros

- ✓ easy to reproduce; low-cost; disposable
- ✓ Single spherical electrode; regular de-charging
- ✓ Indirect measurement (can provide a better estimation of E-field)
- ✓ Sensitivity $\pm 2.3 \text{ mV per V m}^{-1}$; Resolution $\pm 2.3 \text{ mV}$; Accuracy $<0.5\%$

Cons

- × Induced + particle charges
- × Extensive calibration periods;

$$\rho = \frac{\epsilon_0}{d_{eff} w} \frac{1}{dt} \frac{dV_{out}}{dt}$$

Calibration tests (completed)

- Hard vibration test,
- Temperature resilience,
- FW response vs commercial Fieldmill,
- Faraday cage response,
- parallel plates calibration &
- test radiosonde flight

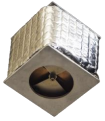
❖ Tethered to MiniMill for balloon-borne launches



Convergence of E-field measurements



MiniMill and Charge sensor parallel post-processing



Correction to the MiniMill E-field for the **rotational position** of the sensor:

$$E_z^{cor} = E_z^{meas} \cos \theta_{roll} \cos \theta_{pitch}$$

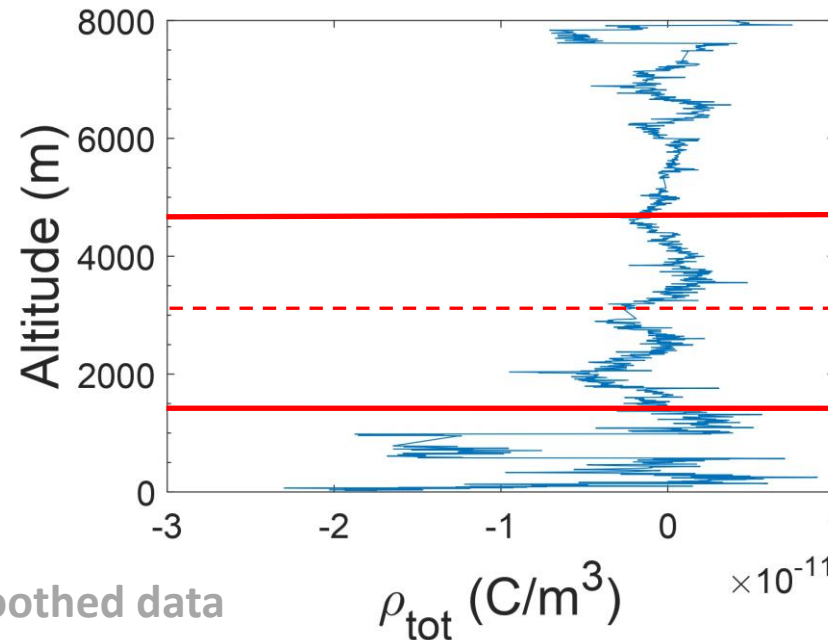
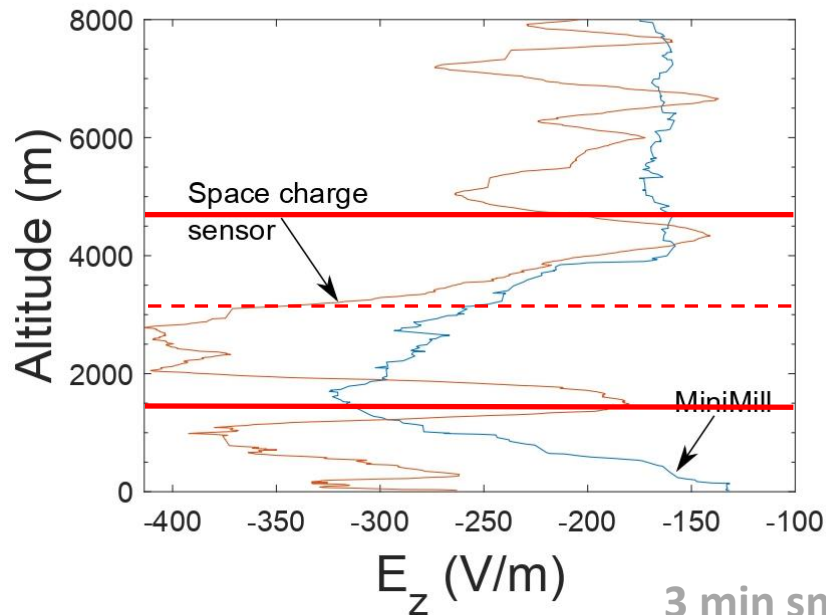
✓ Good agreement

Solving **Poisson** for the charge sensor measurement subsections (with SOR):



$$\frac{d^2V}{dz^2} = -\frac{\rho_{tot}}{\epsilon_0} \quad \text{assuming} \quad E_z = -\frac{dV}{dz}$$

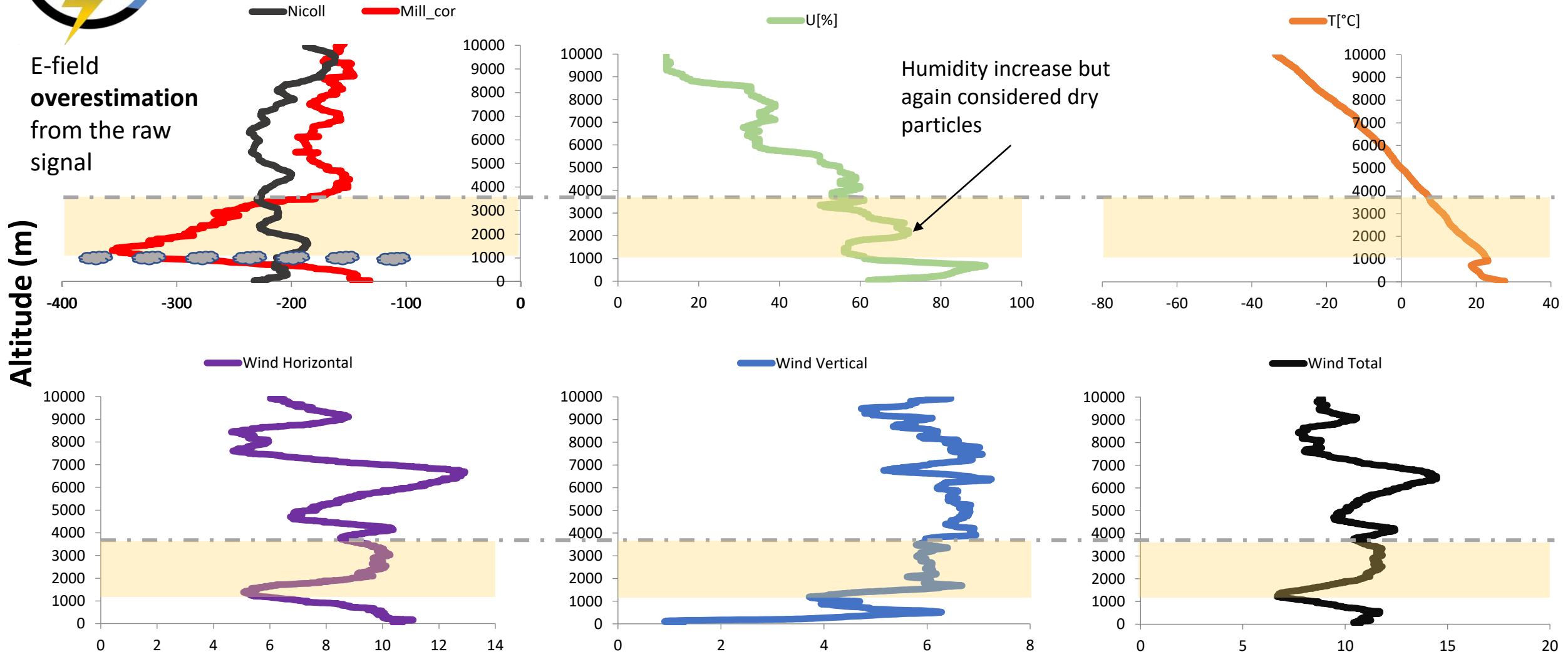
✓ Layer Stratification



✓ Charge sensor's better



Electrical Profiling comparison to meteorological parameters





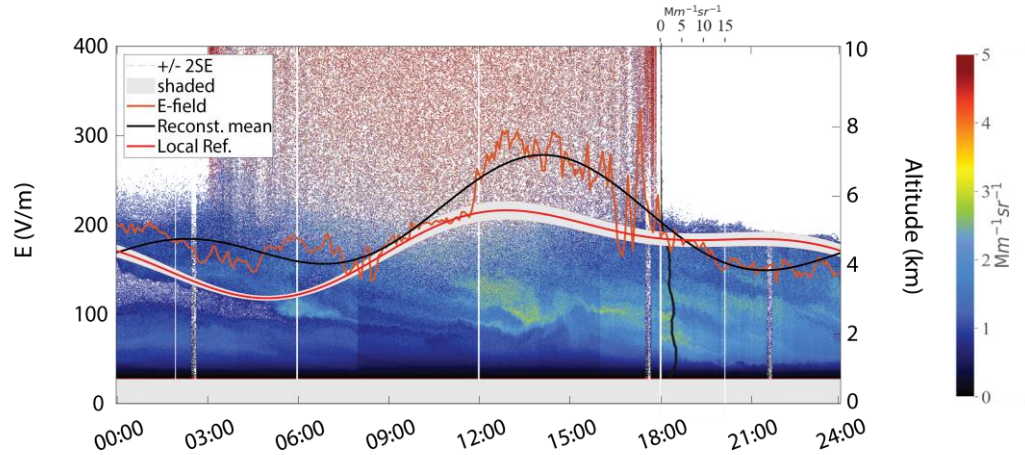
Electrified dust layers detection with Ground-based Methods



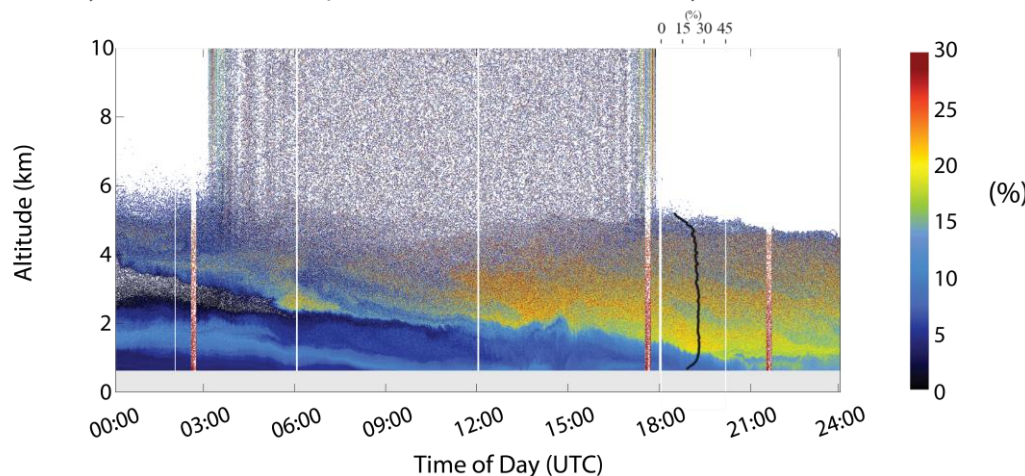
✓ First ever synergistic measurements between a lidar and a ground-based fieldmill

Electrically active

Electric Field strength vs Attenuated Backscatter Coefficient at 532nm - July 25th 2017, Finokalia

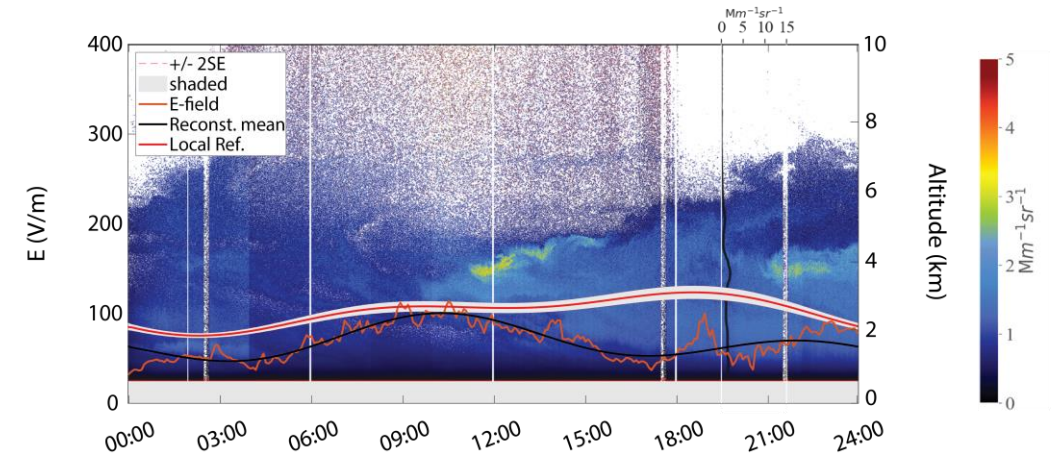


Polly^{XT} Volume Linear Depolarization Ratio at 532nm - July 25th 2017, Finokalia



Electrically inactive or **small charge separation**

Electric Field strength vs Attenuated Backscatter Coefficient at 532nm - June 23rd 2019, Antikythera



Polly^{XT} Volume Linear Depolarization Ratio at 532nm - June 23rd 2019, Antikythera

